

MINOR CHANGE APPROVAL

10069359

This Certificate/Approval is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

AVEO ENGINEERING GROUP S.R.O.

**OBORY C.P. 98
263 01 DOBRIS
CZECH REPUBLIC**

EASA.AP429

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and, if applicable, environmental protection requirements when operated within the conditions and limitations specified below:

Type Certificate Number: EASA.R.010

Type Certificate Holder: AIRBUS HELICOPTERS DEUTSCHLAND

Type: MBB-BK 117

Model: MBB-BK117 C-2

MBB-BK117 D-2

Description of Design Change:

Installation/Replacement of Search/Landing Lights

EASA Certification Basis:

The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:

the following paragraph(s) at a later amendment: CS-29, Amdt. 6
301, 603, 609, 610, 773, 1301, 1309, 1351, 1383, 1529

The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

See Continuation Sheet(s)

For the European Aviation Safety Agency

Cologne, Germany, 21 March 2019



Associated Technical Documentation:

-Installation Instruction (including ICA) AVE-MOD-023-INS, Issue 2
or later revisions of the above listed document(s) approved/accepted under the EASA system.

Limitations/Conditions:

Limitations:

NVIS Operation is not covered by this approval.

Conditions:

AVEO Engineering Group S.R.O. light units with part numbers AVE-LS400KWI-M01, AVE-OLSIOW-M01 and AVR-SISILOW-M01 are included in this approval. They can be installed as replacement or instead of the equivalent parts approved with the original helicopter.

The approval holder shall fulfill the obligations of Part 21, Point 21A.109.

Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -