

SUPPLEMENTAL TYPE CERTIFICATE

10061933 REV. 1

This Certificate/Approval is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

AVEO ENGINEERING GROUP S.R.O.

**OBORY C.P. 98
263 01 DOBRIS
CZECH REPUBLIC**

EASA.AP429

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and, if applicable, environmental protection requirements when operated within the conditions and limitations specified below:

Type Certificate Number: EASA.IM.A.120

Type Certificate Holder: THE BOEING COMPANY

Type: 737

Model: 737-300, 737-400, 737-500

737-600, 737-700, 737-800

737-900, 737-900ER

Description of Design Change:

Design Change for Boeing 737 Light replacement.

Replacement of existing Navigation-, Logo-, Wing Illumination-, Taxi- or Turn lights with AVEO Engineering LED lights.

Revision 1 is the addition of replacement of light bulbs for strobe lights, the anti-collision lights and the landing light and replace the previously approved Wing Illumination and Logo light with next generation LED light.

See Continuation Sheet(s)

For the European Aviation Safety Agency

Cologne, Germany, 07 November 2018



**Supplemental Type Certificates & Special Projects
Section Manager**



EASA Certification Basis:

The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:

The following paragraph(s) at a later amendment: CS 25 AMDT 18: 25.301; 25.603; 25.609; 25.773; 25.1301; 25.1309; 25.1351; 25.1385; 25.1387; 25.1389; 25.1391; 25.1393; 25.1395; 25.1397; 25.1403; 25.1529.

The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

Associated Technical Documentation:

The change is defined with Issue 05 of the Master Documents List AVE-MOD-002-MDL, or later revisions of the above listed document(s) approved/accepted under the EASA system.

Limitations/Conditions:

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -