

MINOR CHANGE APPROVAL

10064674

This Minor Change Approval is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

AVEO ENGINEERING GROUP S.R.O.

**OBORY C.P. 98
263 01 DOBRIS
CZECH REPUBLIC**

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and environmental protection requirements when operated within the conditions and limitations specified below:

Original Type Certificate Number: EASA.IM.A.191

Type Certificate Holder: BOMBARDIER INC

Type: DHC-8

Model: DHC-8-301

DHC-8-311

DHC-8-314

DHC-8-315

DHC-8-401

DHC-8-402

Description of Design Change:

Cabin Light LED replacement

This modification is the replacement of the fluorescents lights of the cabin side wall and the ceiling, both installed in the overhead bins

See Continuation Sheet(s)

For the European Aviation Safety Agency

Cologne, Germany, 15 February 2018



Michael SINGER
Section Manager
Business Aeroplanes



EASA Certification Basis:

The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:

the following paragraph(s) at a later amendment: Amendment 20 of CS 25.301; CS25.603; CS25.609; CS25.1301; CS25.1309; CS25.1351; CS25.1529.

The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

Associated Technical Documentation:

- AVE-MOD-015-INS; Installation Instructions, Issue 01
- AVE-MOD-015-MCS; Modification Conformity Statement
- F-AVE-001A-A, Occurrence Report Form

Limitations/Conditions:

The approval holder shall fulfill the obligations of Part 21, Paragraph 21A.109 or 21.A.451(b) (as applicable).

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -