

MINOR CHANGE APPROVAL

10070477

This Certificate/Approval is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

AVEO ENGINEERING GROUP S.R.O.

**OBORY C.P. 98
263 01 DOBRIS
CZECH REPUBLIC**

EASA.AP429

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and, if applicable, environmental protection requirements when operated within the conditions and limitations specified below:

Type Certificate Number: EASA.R.508

Type Certificate Holder: AIRBUS HELICOPTERS

Type: EC 120

Model: EC 120 B

Description of Design Change:

Change of external Lights to LED

EASA Certification Basis:

The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:

The following paragraph(s) at a later amendment: CS-27, Amdt. 6 - 301, 603, 609, 610, 773, 1301, 1309, 1351, 1383, 1385, 1387, 1389, 1391, 1393, 1395, 1397, 1401, 1529.

The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

See Continuation Sheet(s)

For the European Aviation Safety Agency

Cologne, Germany, 12 July 2019



Volker ARNSMEIER
Section Manager
eVTOL & New Concepts



Associated Technical Documentation:

- Installation Instruction (including CAW requirements) AVE-MOD-030-INS, Iss. 02
 - Drawing List AVE-MOD-030-DL, Iss. 01
 - Flight Manual Supplement AVE-MOD-030-RFM, Iss. 01
- or later revisions of the above listed document(s) approved/accepted under the EASA system.

Limitations/Conditions:

The approval holder shall fulfill the obligations of Part 21, Point 21.A.109.

Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -