

MINOR CHANGE APPROVAL

10077290

This Certificate/Approval is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

AVEO ENGINEERING GROUP S.R.O.

OBORY C.P. 98 263 01 DOBRIS CZECH REPUBLIC

EASA.AP429

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and, if applicable, environmental protection requirements when operated within the conditions and limitations specified below:

Type Certificate Number: EASA.IM.A.503

US 3A20

Type Certificate Holder: TEXTRON AVIATION INC.

Type: Beechcraft 65, 70, 90 Model: B90, C90, E90, H90 C90A, C90GT, C90GTI

Description of Design Change:

Replacement of position lights (type I, II & III), anti-collision lights, strobe lights, taxi lights and the landing lights by LED type lights.

EASA Certification Basis:

The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any impact on the noise database.

Associated Technical Documentation:

Aveo Engineering Group Installation Instructions AVE-MOD-037-INS Issue 01, dated 31 August 2021

See Continuation Sheet(s)

For the European Union Aviation Safety Agency

Cologne, Germany, 16 September 2021

Manfred REICHEL Section Manager

Small Aircraft, Balloons & Airships Section



Task Number: 60079940

AVEO ENGINEERING GROUP S.R.O. - 307368



Aveo Engineering Group AMM Supplement AVE-MOD-037-SUP-AMM Issue 01, dated 3 September 2021 or later revisions of the above listed document(s) approved/accepted under the EASA system.

Limitations/Conditions:

The approval holder shall fulfill the obligations of Part 21, Point 21A.109.

Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

