



INSTALLATION MANUAL

AVE-AXT-IM

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Part 0 Document Administration

0.1 Document Approval

This document has been established in accordance with an alternative procedure to DOA approved under EASA AP429.

This installation manual is applicable for following part numbers:

 Atlas NXT 	- AVE-AXTTLFW-D01
 Atlas NXT Custom 	- AVE-AXTTLFW-D02
 Atlas NXT – Q holder 	- AVE-AXTTLFW-D03
 Atlas NXT – O holder 	- AVE-AXTTLFW-D0 <mark>4</mark>
 Atlas NXT Custom 	- AVE-AXTTLFW-D05

	P.	
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Petr Jaroš Engineer, Aveo Engineering Group, s.r.o.

Approved by:

_____ 14 Oct 2024

Georg Hartl Head of DO, Aveo Engineering Group, s.r.o.



0.2 Amendment Record Procedure

The master copy of this document shall be kept electronically as a read only document under the control of Aveo Engineering Group, s.r.o. as Master Copy.

ALL amendments to this manual will initiate a raise of issue.

The original issue will be identified by **"01"**, and subsequent issues will be numbered sequentially from 02 to 99 in Table 01 - *Issue No.* column.

ALL issues of this document will be approved by Head of DO.

Issue No.	Details	Date of issue	Affected Pages		
01	Initial Issue	17 Oct 2022	ALL		
02	Added new Part Numbers Technical specification update Optic simulation addition Caution note addition	14 Oct 2024	<mark>3-5</mark> 7-10 15-17 19		
Table 01: Record of Document Amendments					

0.3 Affected Pages Procedure

ALL pages affected by ANY raise of issue of this document will be listed in Table 01 - *Affected Pages* Column.

The reason(s) for **EACH** raise of issue and the description of respective change will be provided in Table 01 - **Details** Column.

Changes from the previous issue are shown as follows:

- a) new text is highlighted with yellow shading: new
- b) deleted text is shown with yellow shading and a strike through: deleted



Part 1 Installation data

1.1 Product Info

Atlas NXT^m is a high powered Landing and Taxi LED light used in GA and transport category aircraft. It has been designed to be lightweight and with a low power draw to meet the highest requirements of all certified aircraft.

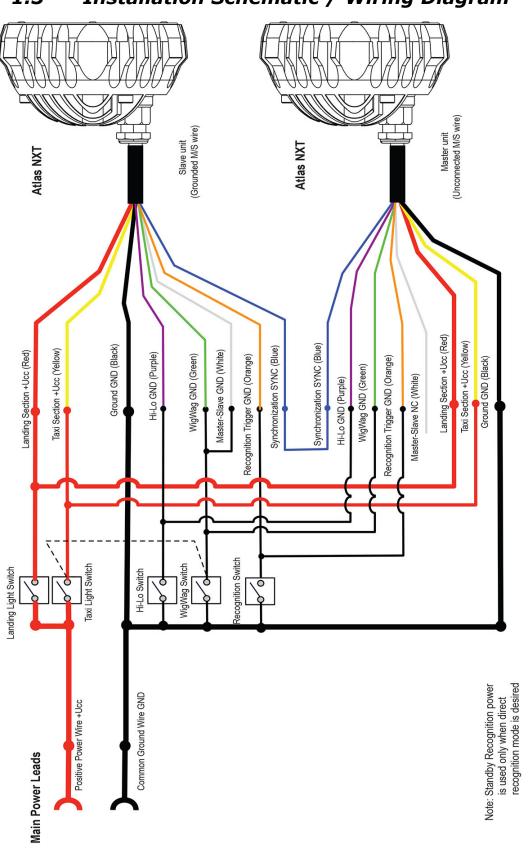
• Atlas NXT	- AVE-AXTTLFW-D01
Atlas NXT Custom	- AVE-AXTTLFW-D02
 Atlas NXT – Q holder 	- AVE-AXTTLFW-D03
 Atlas NXT – O holder 	- AVE-AXTTLFW-D0 <mark>4</mark>
Atlas NXT Custom	- AVE-AXTTLFW-D05

1.2 Operating Instructions

When installed on the aircraft, using the aircraft's power (14 volts or 28 volts), the light will be at its maximum intensity.

Operating Voltage range is +9...+36VDC.





1.3 Installation Schematic / Wiring Diagram



1.4 Control & Power Inputs

Length: 300 mm / 11.8"

1.5 Technical Specification

Light characteristics: Voltage range: Voltage protection: Performance:		Multimode Light (Landing, Taxi, WigWag, Hi/Lo) 9-36VDC a. Transcient voltage: 2 seconds +80VDC b. Over-voltage lockout: +36VDC, not less		onds +80VDC	
Landing:	•	current: power:	2.5A @14VD0 35W @14VD0	•	1.3A @28VDC 36.4W @28VDC
Taxi:	•	current: power:	1A @14VDC, 14W @14VD0		0.55A @28VDC 15.4W @28VDC
Chromaticity: Ambient temperature: Overheat protection: Wiring: Useful life:		Cool White -55°C+85°C / -67°F+185°F Yes See section 1.3 not less than 15.000 aircraft flight hours			
Weight: AVE-AXTTLFW-D01 AVE-AXTTLFW-D02 AVE-AXTTLFW-D03 AVE-AXTTLFW-D04 AVE-AXTTLFW-D05		323 g / 0.71 325 g / 0.72 344 g / 0.76 359 g / 0.79 325 g / 0.72	lb lb lb		



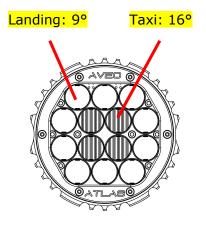
Viewing Angle:

AVE-AXTTLFW-D01 AVE-AXTTLFW-D02 AVE-AXTTLFW-D03 AVE-AXTTLFW-D04

Landing: 9°

Taxi: 12° x 30°

AVE-AXTTLFW-D05



DO160G Test qualified:

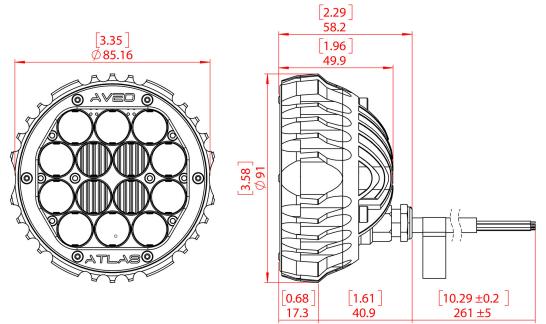
Environment	Section	Category
Temperature / Altitude	4	F2
Temperature Variation	5	A
Humidity	6	С
Operational Shock and Crash Safety	7	B*
Vibration	8	U, curves G
Explosive Atmosphere	9	Н
Waterproofness	10	R
Fluids Susceptibility	11	F**
Sand and dust	12	D
Fungus	13	F
Salt Spray	14	Т
Magnetics Effects	15	Z
Power Input	16	Z
Voltage Spike	17	A
Audio Freq. Conducted Susceptibility	18	Z
Induced Signal Susceptibility	19	ZC
Radiated and Conducted Susceptibility	20	Π
Radiated and Conducted Emissions	21	В
Icing	24	A and C

* Aircraft Type: 5. Helicopters and all Fixed-Wing Test Type R, 20.0g all direction **Actual fluids: Jet A-1 aviation fuel, Mobil Jet Oil II, Ethylene glycol de-icing fluid

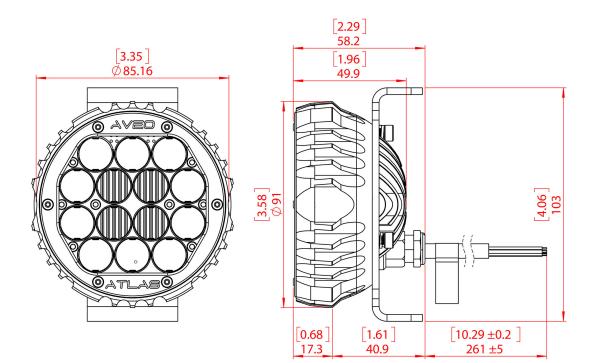


1.6 Technical Drawing

AVE-AXTTLFW-D01

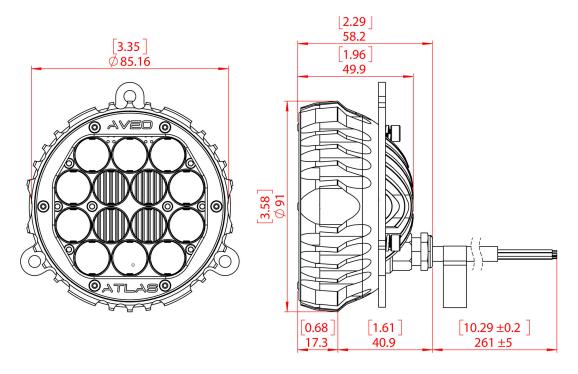


AVE-AXTTLFW-D04

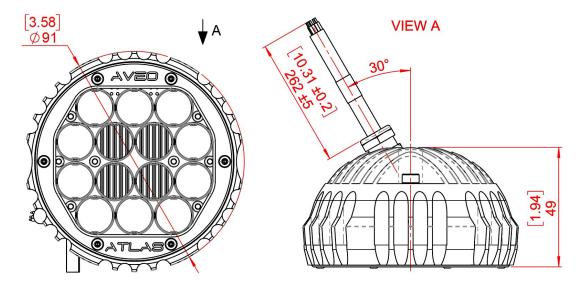


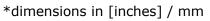


AVE-AXTTLFW-D03





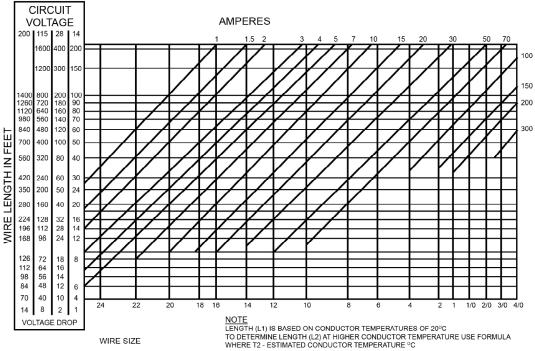






1.7 Wiring Chart

Use diagram below defining the wiring size depending on the current and the wire length. Make sure you add up the current for all connected lights. If current is not given, then divide the power by the voltage.



VOLTAGE DROP CHART INTERMITTENT FLOW AT 20° TIN-PLATED MIL-W-27759 CONDUCTOR



1.8 Optic Simulation

AVE-AXTTLFW-D01, AVE-AXTTLFW-D02, AVE-AXTTLFW-D03, AVE-AXTTLFW-D04

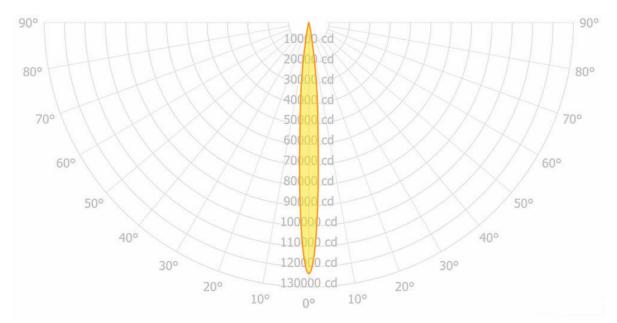
Landing Hi

ENGINEERING

GROUP

123 000 cd

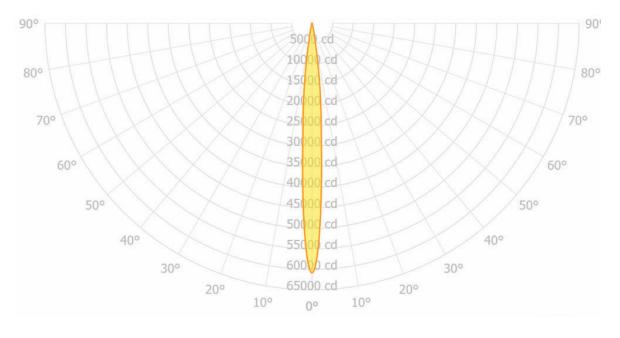
8.3° (at 50% of max light intensity) 18° (at 10% of max light intensity)



<u>Landing Lo</u>

60 800 cd

8.6° (at 50% of max light intensity) 18° (at 10% of max light intensity)

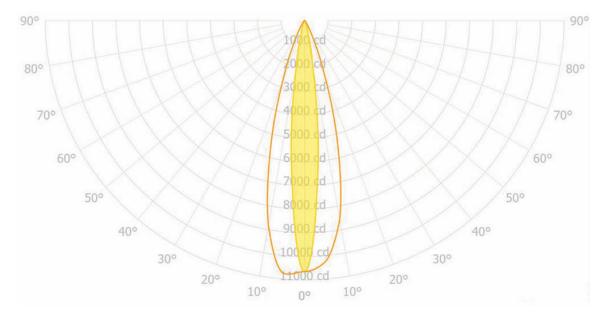




<u>Taxi Hi</u>

10 700 cd

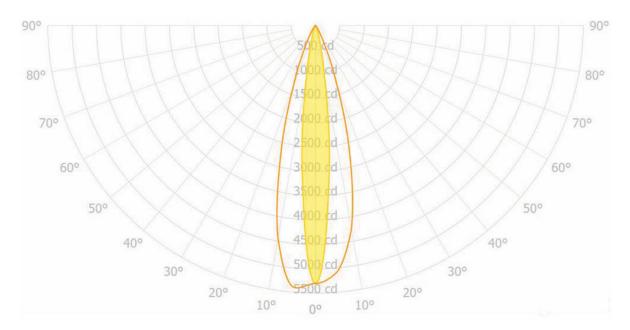
12° x 30° (at 50% of max light intensity) 26° x 52° (at 10% of max light intensity)



<u>Taxi Lo</u>

5 390 cd

12° x 30° (at 50% of max light intensity) 26° x 52° (at 10% of max light intensity)

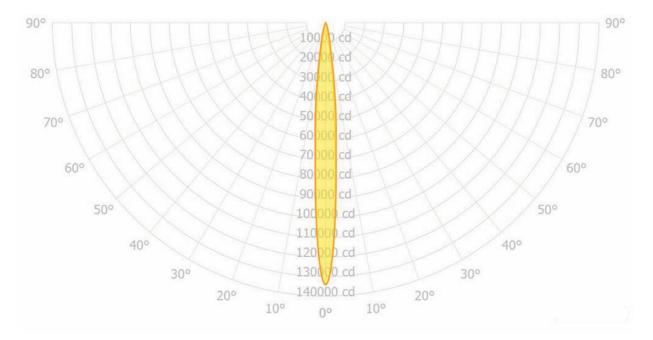




Landing and Taxi sections Hi

134 000 cd

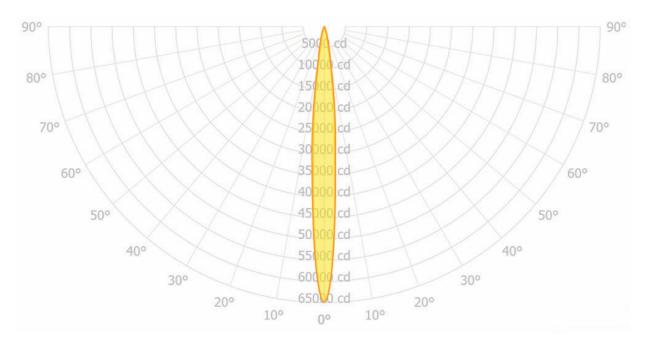
8.5° x 9.2° (at 50% of max light intensity) 19° x 22° (at 10% of max light intensity)



Landing and Taxi sections Lo

64 900 cd

8.7° x 9.7° (at 50% of max light intensity) 20° x 24° (at 10% of max light intensity)

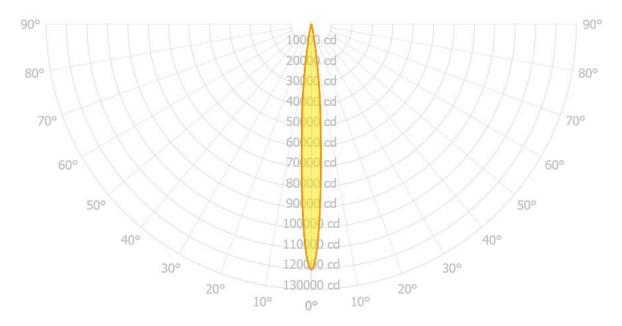




<u>Landing Hi</u>

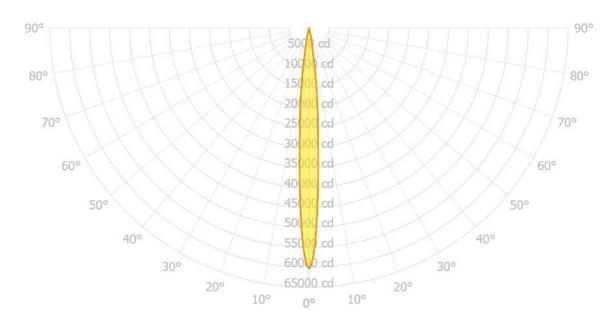
121 000 cd

8.5° (at 50% of max light intensity) 18° (at 10% of max light intensity)



Landing Lo

60 000 cd



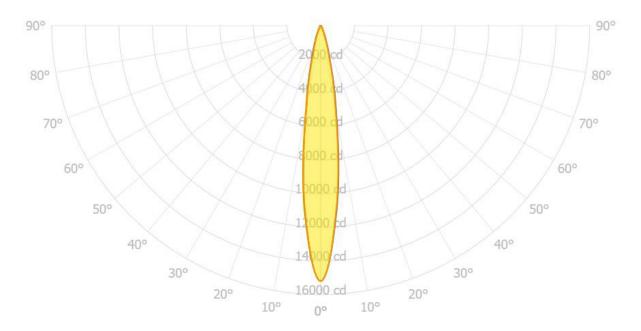
8.5° (at 50% of max light intensity) 18° (at 10% of max light intensity)



<u>Taxi Hi</u>

15 100 cd

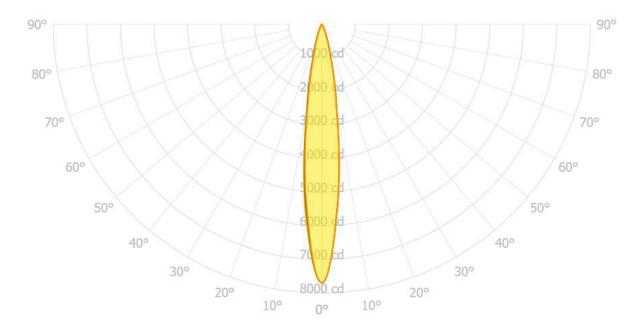
15.8° (at 50% of max light intensity) 34° (at 10% of max light intensity)



<u>Taxi Lo</u>

7 600 cd

15.8° (at 50% of max light intensity) 34° (at 10% of max light intensity)



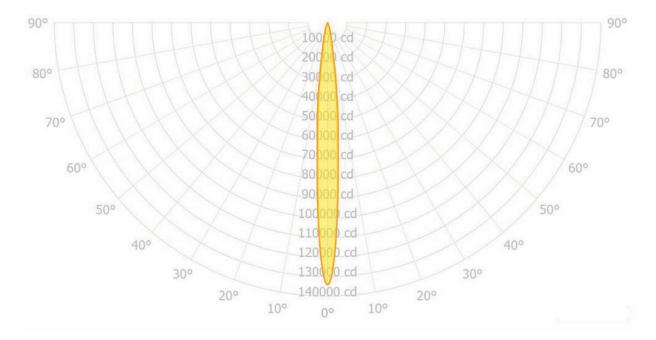




Landing and Taxi sections Hi

136 000 cd

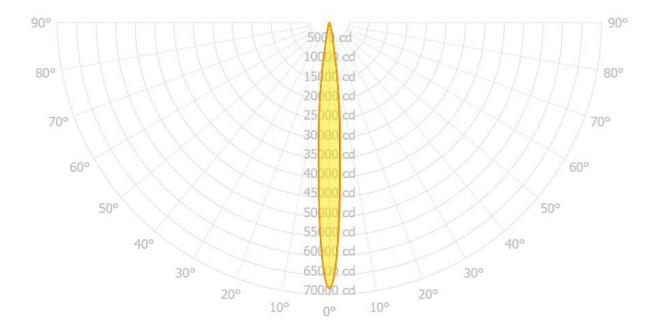
9.3° (at 50% of max light intensity) 20° (at 10% of max light intensity)



Landing and Taxi sections Lo

68 000 cd

9.3° (at 50% of max light intensity) 20° (at 10% of max light intensity)





1.9 Equipment Limitation

Atlas NXT should only be powered by 9-36VDC

1.10 Care and Cleaning of Lights

Aveo Engineering Aviation Lights are factory polished and delivered as ready to install on the aircraft.

If the lights need a deeper cleaning, they should be polished with a quality lamb's wool sheet that is suitable also for deeper polishing. Under no circumstances should any petroleum based product be used to clean the lights.

1.11 Testing the Lights before Installation

All Aveo Aviation lights undergo rigorous testing prior to being released from our engineering manufacturing department. This testing involves a burn-in time as well as other function testing. No light is released for sale without undergoing this extensive operational testing.

When you receive the **Atlas NXT** light, and wish to test the function of the light prior to installation on your aircraft, please note the following:

1. Please review the written information that is enclosed in the packaging. Warranty information as well as a cautionary note about power supply removal is enclosed with each package.

2. Remove the light from the package. Note that there are 6 wires:

RED	Landing (AWG20)
YELLOW	Taxi (AWG20)
BLACK	GND (AWG20)
GREEN/YELLOW	Chassis (AWG20)
WHITE	Master/Slave (AWG24)
BLUE	Synchro (AWG24)
VIOLET	Hi/Lo (AWG24)
GREEN	WigWag (AWG24)

3. Testing of the function of the light can be done with a regular 14VDC or 28VDC power supply (not a battery charger). Connect the ground wire to black wire and then connect the power wire to the red wire. The landing light section should start lighting. Disconnect the power wire. Connect the power wire to the yellow wire. The taxi light section should start lighting. When installed on the aircraft, using the aircraft's power (14VDC or 28VDC), the light will be at its maximum intensity.

After testing, the light can be installed on the aircraft.





IMPORTANT NOTES:

Under no circumstances should any power supply other than a 9-36VDC, or a 14 or 28 Volt battery be used to test the light. Do not use: Battery chargers, battery back-up power devices, or other bench avionics testing methods to test the aviation light. The light is functional between 18-36 volts. Use of a battery charger or other power unit to test the light will void the warranty and may damage the light.

If you have any questions about the installation of the lights, please refer to our web site: <u>www.aveoengineering.com</u>

1.12 Continued Airworthiness Information

Circuit/Wiring Protection

Each Titan series light features a **Negative Temperature Coefficient** (NTC) circuit that limits internal temperatures by attenuating operating current (with a corresponding reduction of brightness) when internal temperatures reach a certain threshold. This proprietary circuitry serves to protect the light itself, and associated aircraft wiring, against a thermal runaway condition.

Periodic Inspection Procedure for Atlas NXT

The **Atlas NXT** light should always be checked for proper operation during preflight. This procedural information is already provided in all general aviation aircraft flight manuals.

The lights should be visually inspected for general condition, proper operation, and correct installation at each annual and/or 100 hours inspection. In addition refer to section 1.10 of installation manual for detailed cleaning instructions.

1.13 RoHS Compliance Statement

Scope

This statement clarifies Aveo Engineering's compliance with European Union Directive 2015/863/EU on the restriction of the use of certain hazardous substances in electrical and electronic equipment ("RoHS") that took effect on June 4, 2015. The RoHS Directive restricts the sale of electronic equipment containing certain hazardous substances in the European Union including:



Cadmium(Cd): 0.01% Mercury: 0.1% Lead(Pb) : 0.1% Hexavalent chromium (Cr6+) : 0.1% Polybrominated biphenyls (PBB): 0.1 %; Polybrominated diphenyl ethers (PBDE): 0.1 % Bis(2-Ethylhexyl) phthalate (DEHP): 0.1% (added in 2015); Benzyl butyl phthalate (BBP): 0.1% (added in 2015); Dibutyl phthalate (DBP): 0.1% (added in 2015); Disobutyl phthalate (DIBP): 0.1% (added in 2015)

Compliance

Aveo Engineering certifies that all products sourced from manufacturing facilities comply with the environmental standards set forth by the Directive 2015/863/EU, recast amendment of RoHS Directive 2011/65/EU Article (4), and do not contain any of the above-mentioned, 10 hazardous substances above the specified limits. All products manufactured by Aveo Engineering are RoHS-compliant. With regards to RoHS-2 CE marking, product packaging is labeled attesting conformity if required.

References

Directive 2015/863/EU of the European Parliament and of the Council on the restriction of the use of certain hazardous substances in electrical and electronic equipment.

1.14 EU REACH Regulation (EC) No. 1907/2006

Aveo Engineering declares that no chemicals are produced and that none of the chemicals used during the production process or needed for the product maintenance or service, is listed on the current European Chemicals Agency's Candidate list of Substances of Very High Concern for Authorization.